

Overview of the New NOVAS (Fortran)

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The 2004 version of Naval Observatory Vector Astrometry Subroutines (NOVAS version F2.9, in Fortran) implements the resolutions on astronomical reference systems and Earth rotation models passed at the IAU General Assemblies in 1997 and 2000. Although the IAU resolutions provided basic guidance, many of the models and parameter values needed to implement the resolutions in practice did not become available until late 2002. The International Earth Rotation and Reference Systems Service (IERS), which had been the driving force for many of the resolutions, implemented these models for their own data processing at the beginning of 2003. (A dynamically consistent precession theory is still in preparation. The precession in NOVAS version F2.9 is a preliminary algorithm adopted by the IERS. The final precession will go into NOVAS version F3.0.) This version of NOVAS also improves the accuracy of its star and planet position calculations (apparent places) by including several small effects not previously implemented in the code. A number of new convenience functions have also been added.

A detailed list of the changes in NOVAS from the previous version (F2.0 of 1998) is given in the Appendix. The following paragraphs are meant to provide some perspective for people who are already familiar with NOVAS. To the greatest extent possible, the calling sequences for the highest-level (and most used) functions from the previous versions of NOVAS have been preserved — but there are a few important exceptions. There are many new calls.

Important Changes in Existing Calls

Probably the most important change to existing NOVAS calls is the change of proper motion and parallax units in the calls to APSTAR, VPSTAR, and ASSTAR, CATRAN, and GETHIP, all of which deal with star positions. The units have been changed as follows:

proper motion in right ascension: from seconds of RA per century to milliarcseconds per year
proper motion in declination: from arcseconds per century to milliarcseconds per year
parallax: from arcseconds to milliarcseconds

These changes have been made to conform to the units used in most modern star catalogs (e.g., Hipparcos, Tycho-2, or the FK6), which in turn follow from the observational techniques now used in the construction of such catalogs. Obviously, star data previously used with NOVAS must either be replaced or transformed. The transformation equations from “old” to “new” units are as follows:

```
PMRNEW = PMROLD * 150.D0 * DCOS ( DEC0 * DEGRAD ) ; proper motion in RA
PMDNEW = PMDOLD * 10.D0 ; proper motion in dec
PAXNEW = PAXOLD * 1000.D0 ; parallax
```

where `DEC0` is the catalog declination (J2000.0 or ICRS) of the star in degrees and `DEGRAD` is the degrees-to-radians conversion factor (0.01745329...).

The other major change to a high-level subroutine is that `PNSW` has been renamed to `TERCEL` (it carries out the terrestrial-to-celestial transformation), with a change to the time argument. All other

changes to existing NOVAS calls involve lower-level routines not frequently invoked by most users; these are detailed in the Appendix.

PLACE: A New General-Purpose “Place” Subroutine

All computational code to compute apparent, topocentric, virtual, astrometric, etc., places of stars or planets has now been consolidated into a single new subroutine call PLACE. The familiar calls to APSTAR, APPLAN, TPSTAR, etc., are now just “front-ends” to PLACE. This has eliminated much duplicate code and also provides more flexibility and possible future additions (such as binary star orbits or nonlinear terms in proper motion). PLACE can also provide star or planet positions within the “intermediate” coordinate system that is part of the new paradigm for Earth rotation calculations (see below). PLACE provides its output position both in spherical coordinates (right ascension, declination, and, for solar system bodies, geometric distance) and as a unit vector. PLACE accepts the specification of solar system bodies by name, e.g., ‘MARS’, ‘SATURN’, or ‘SUN’, thus increasing the readability of code. You may want to consider changing your calls to APSTAR, APPLAN, etc., to the equivalent calls to PLACE.

The International Celestial Reference System (ICRS)

Reference data for positional astronomy, such as the data in astrometric star catalogs (e.g., Hipparcos) or barycentric planetary ephemerides (e.g., JPL’s DE405) are now specified within the International Celestial Reference System (ICRS). The ICRS is a coordinate system whose origin is at the solar system barycenter and whose axis directions are effectively defined by the adopted coordinates of about 600 extragalactic radio sources observed by VLBI (see Section H of *The Astronomical Almanac*). These radio sources (quasars and active galactic nuclei) are assumed to have no observable intrinsic angular motions. Thus, the ICRS is a “space-fixed” system (more precisely, a kinematically non-rotating system) without an associated epoch. However, the ICRS closely matches the conventional dynamical system defined by the Earth’s mean equator and equinox of J2000.0; the alignment difference is at the 0.02 arcsecond level, negligible for many applications.

NOVAS now assumes that input reference data, such as catalog star positions and proper motions, and the basic solar system ephemerides, are provided in the ICRS, *or at least are consistent with it to within the data’s inherent accuracy*. The latter case will probably apply to most FK5-compatible data specified in the dynamical system of J2000.0. The distinction between the ICRS and the dynamical system of J2000.0 becomes important only when an accuracy of 0.02 arcsecond or better is important. Nevertheless, because NOVAS is designed for the highest accuracy applications, you will now see the ICRS mentioned as the reference system of choice for many input arguments to NOVAS subroutines.

NOVAS now contains a subroutine called FRAME that transforms vectors from the ICRS to the dynamical system of J2000.0 or vice versa. This transformation is a very small fixed rotation. FRAME is called many times, in both directions, within the NOVAS code. That is because precession (and nutation) can properly be applied only to vectors in a dynamical system; vectors in the ICRS must be transformed, via FRAME, to the dynamical system (equator and equinox) of J2000.0 before PRECES is used. If your code only interacts with the highest level NOVAS subroutines, all this is transparent to you. However, if you use PRECES within your own code, you should precede it by a call to FRAME (with the middle argument $K>0$) if your input vector is expressed in the ICRS.

New Features

Subroutines have been added to NOVAS that provide new functionality and convenience:

PLACE: A new general-purpose apparent place subroutine (see paragraph above).

IDSS: An integer function that returns a solar system body's identification number (which is used in various NOVAS subroutine calls), given the body's name as a character string. For example, `IDSS('MARS')` usually equals 4. Because IDSS is a function, it can be referred to within calls to other NOVAS subroutines, e.g.,

```
CALL APPLAN ( TTJD, IDSS('JUPITER'), IDSS('EARTH'), RAJUP, DECJUP, DISJUP )
```

(If you supply your own version of subroutine SOLSYS, you must also now supply a corresponding version of IDSS.)

GETVEC: A subroutine that returns the last NOVAS-computed celestial position (apparent or astrometric place, etc.) as a unit vector. The vector is expressed in the same reference system as the previously supplied spherical coordinates.

EQECL: Converts right ascension and declination to ecliptic longitude and latitude.

EQGAL: Converts ICRS right ascension and declination to galactic longitude and latitude.

ICRSEQ: Converts ICRS right ascension and declination to one of the equatorial systems of date.

ASTCON: Supplies the value of an astronomical constant, given its name as a character string. The values of all fundamental astronomical constants used by NOVAS are stored within this subroutine and nowhere else. The names of the constants available and the units used for each are listed in the subroutine's preamble. For example, `CALL ASTCON ('ERAD', 1.D0, RADIUS)` returns, in argument `RADIUS`, the value of the equatorial radius of the Earth in meters.

New Models for Precession and Nutation

It has been known for over a decade that the standard models for the precession and nutation of the Earth's rotation axis have been in need of revision. The value of the angular rate of precession in longitude adopted by the IAU in 1976 — and incorporated into the widely used Lieske et al. (1977) precession formulation — is too large by about 0.3 arcsecond per century (3 mas/yr). The amplitudes of a number of the largest nutation components specified in the 1980 IAU Theory of Nutation are also known to be in error by several milliarcseconds. Both the precession and nutation errors are significant relative to current observational capabilities.

Thus, the resolutions passed by the IAU in 2000 mandated an improvement to the precession and nutation formulations. The new version of NOVAS incorporates the models adopted by the IERS in response to these resolutions. (As mentioned above, the precession model is still a work in progress, and an IAU Working Group is in the process of finalizing a dynamically consistent model. NOVAS version F2.9 contains a preliminary IERS formulation of precession and NOVAS version F3.0 will contain the final IAU model. The difference in results, for dates within several decades of the year 2000, will be negligible for practical purposes.) Despite the new models, from a programming point of view, the subroutines that directly involve precession and nutation — `PRECES`, `NUTATE`, `ETILT`, `NOD`, and `SIDTIM` — work the same as before, but with slightly different results. It should be noted

that the new nutation model has more than ten times the number of trigonometric terms than the previous model. Since evaluation of nutation has always been the most computationally intensive task in NOVAS, you may notice an increase in execution time for some NOVAS applications (more on this below).

A New Model for the Rotation of the Earth about its Axis

IAU resolutions passed in 2000 also dealt in a very fundamental way with how we describe the Earth's spin *around* its axis. The conventional treatment is based on the equinox and sidereal time; Greenwich (or local) sidereal time is just the Greenwich (or local) hour angle of the equinox of date. However, the equinox is constantly moving due to precession, so that sidereal time combines two angular motions, the Earth's rotation and the precession of its axis. (In the case of apparent sidereal time, nutation is also mixed in.) One rotation of the Earth is about 0.008 second longer than one mean sidereal day.

For about two decades, some of the people who routinely deal with the most precise measurements of the Earth's rotation have been advocating for a change in the way it is described, and their ideas were introduced in resolutions passed by the IAU in 2000. In this new paradigm, the reference point on the moving celestial equator for the description of Earth rotation is called the celestial ephemeris origin (CEO). Unlike the equinox, this point has no motion along the equator at all; as the orientation of the equator changes in space due to precession and nutation, the CEO remains on the equator but its instantaneous motion is always at right angles to it. Thus, loosely speaking, two transits of the CEO across the local meridian define one rotation of the Earth. The CEO is a point on the celestial equator near RA=0, and there is a corresponding point on the terrestrial equator near longitude=0 called the terrestrial ephemeris origin (TEO). For all astronomical purposes, the TEO can be considered a point fixed on the surface of the Earth at latitude and longitude zero.¹ In the new paradigm, the rotation of the Earth is specified by the angle (in the instantaneous equatorial plane) between the TEO and the CEO, which is a linear function of universal time (UT1). This angle is called the Earth rotation angle and is designated by θ .

How are hour angles of celestial objects computed in the old and new paradigms? Assume that we are considering Greenwich hour angles, that is, hour angles measured from the meridian of longitude zero, and without polar motion. In the equinox-based scheme, we compute the topocentric apparent place of the object of interest with respect to the true equator and equinox of date. Then we compute apparent sidereal time and subtract the object's apparent right ascension to form the hour angle. In the CEO-based scheme, we compute the object's topocentric apparent place with respect to the true equator and CEO of date. To form the hour angle, we compute the Earth rotation angle and subtract the equatorial angle measured eastward from the CEO to the object (essentially, the right ascension of the object measured with respect to the CEO). Since hour angle is an observable quantity, the two results should be identical. You might wonder, then, what the advantage of the new system is. In the equinox-based scheme, precession and nutation appear in both the apparent place of the star and sidereal time. In the CEO-based scheme, they appear only in the apparent place of the star. The CEO-based method also does not depend on the equinox, and is thus independent of any model of the Earth's orbital motion.

¹ The CEO and TEO are technically examples of *non-rotating origins*, and neither is fixed within its respective coordinate system. However, the slow drift of the TEO, due to polar motion, with respect to standard geodetic coordinates (the International Terrestrial Reference System, or, effectively, WGS84) amounts to only 1.5 millimeters per century and is completely negligible for astronomical purposes.

The following table summarizes the two equivalent procedures for hour angle and the NOVAS subroutines that would be used for each, assuming that polar motion is neglected. The procedures outlined here provide the Greenwich hour angle of a star.

	Equinox-Based Method	CEO-Based Method
Use subroutine	APSTAR followed by TPSTAR — or — PLACE with OBJECT='STAR', LOCATN=1, and ICOORD=1	PLACE with OBJECT='STAR', LOCATN=1, and ICOORD=2
... to obtain	RA and DEC, the topocentric apparent right ascension and declination of the star with respect to the equator and equinox of date (in hours and degrees, respectively)	RA and DEC, the topocentric apparent right ascension and declination of the star with respect to the equator and CEO of date (in hours and degrees, respectively)
Then use subroutine	SIDTIM with K=1	EROT
... to obtain	GST, Greenwich apparent sidereal time (in hours)	THETA, the Earth rotation angle, θ (in degrees)
Compute Greenwich hour angle	GHA = GST – RA, (in hours)	GHA = THETA / 15.D0 – RA, (in hours)

The computed GHA may have to be reduced to the range -12^h to $+12^h$. Subroutines APSTAR and PLACE require time arguments in the TT time scale, while TPSTAR, SIDTIM, and EROT require time arguments in the UT1 time scale. The two procedures should yield the same value of GHA to within several microarcseconds and identical values for DEC.

Two high-level NOVAS subroutines that involve Earth rotation, SIDTIM and TERCEL (the latter replaces the old PNSW) can actually perform their internal calculations using either the equinox-based paradigm or the CEO-based paradigm.² (Note: ZDAZ is also affected because it calls TERCEL.) The method used is selected by a prior call to either EQINOX or CEOTEO (without arguments), which remains in effect until changed. Since there is no external difference in how SIDTIM or TERCEL are used, and the two computational paradigms yield answers that are consistent within a few microarcseconds over many centuries, there is seldom a practical basis for a choice. However, the equinox method must be used for dates before 1700 or after 2300, and is much more efficient if mean sidereal time is to be computed. The equinox-based paradigm is the default, that is, it is used unless CEOTEO has been called. That will, of course, be the case for any existing programs that are not updated to make this choice explicit.

Another choice is now available that has a more practical effect: Earth rotation calculations can be performed in either high- or low-accuracy mode. A call to either HIACC or LOACC (without arguments) sets the accuracy, which remains in effect until changed. High-accuracy mode is the default, with the various models evaluated at the few-microarcsecond level. For nutation, for example, this means that

² It may seem odd that sidereal time can be computed using the CEO-based paradigm, but all that is needed is the angle between the equinox and the CEO (both of which lie in the equatorial plane), and this is straightforward to compute if we know the location of both points in the ICRS.

a 1365-term trigonometric series is evaluated for each unique date. Neither the models nor current observations are accurate at this level, however, so much of the increased computational burden is unproductive. A call to LOACC sets the Earth rotation computations (and *only* those computations) in NOVAS to an accuracy of 0.1 milliarcsecond. The computation time for these calculations is thereby reduced by about 2/3.

Finally, another of the new Earth-rotation-related subroutines is worth mentioning. For a given TDB date, CEORA provides the right ascension of the CEO with respect to the true equator and equinox of date. With a sign reversal, this quantity is the *equation of the origins*, the direction of the true equinox measured in the equator eastward (+) from the CEO. The equinox and CEO can be considered different right ascension origins on the instantaneous equator, and as such they define separate equatorial systems for the equinox-based and CEO-based paradigms. CEORA therefore provides the angular difference between the origins of these two systems.

Some Terminology

Not surprisingly, the IAU resolutions related to Earth rotation have spawned a lot of new terminology, not all of which has become universally accepted. An IAU Working Group on Nomenclature for Fundamental Astronomy has been appointed to try to sort it all out. The most commonly used terms and abbreviations now appear in comment statements in some of the new NOVAS subroutines, including the preambles where the input and output arguments are described. A brief summary of these terms is therefore in order here. The celestial ephemeris origin (CEO) and terrestrial ephemeris origin (TEO) have already been described; these terms are mentioned specifically in the 2000 IAU resolutions. Another term specifically introduced in those resolutions is the celestial intermediate pole (CIP), which is the celestial pole defined by the new precession and nutation models. The true equator of date is thus a plane orthogonal to the CIP. The coordinate system defined by the true equator of date and the CEO is widely referred to as the intermediate system (or the celestial intermediate system), because it is in a sense midway between the rapidly rotating terrestrial latitude-longitude system and the completely non-rotating ICRS. The right-ascension-like coordinate in the intermediate system (the azimuthal coordinate measured in the equatorial plane eastward from the CEO) will probably be called something like CEO right ascension (or right ascension with respect to the CEO).

How NOVAS Implements the CEO-Based Paradigm

The NOVAS implementation of the CEO-based Earth rotation paradigm for a given date is based on the construction of the intermediate system for that date, using vectors toward the celestial intermediate pole (CIP) and the celestial ephemeris origin (CEO). These two directions define, respectively, the z-axis and x-axis of the intermediate system. The direction toward the CIP in the ICRS can be computed by passing the vector (0,0,1) through subroutines NUTATE, PRECES, and FRAME. Given the direction of the CIP, the only remaining piece of required information is the ICRS right ascension of the CEO for the same date, which is provided by CEORAI. The basis vectors of the intermediate system, with respect to the ICRS, are assembled by CEOBAS. Having these basis vectors available allows NOVAS to easily transform any vector in the ICRS to the intermediate system. The only other quantity used in the CEO-based paradigm is the Earth rotation angle, which is trivial to compute and provided by EROT.

The only tricky part of this process is obtaining the ICRS right ascension of the CEO, which is a unique quantity derived from an integration. CEORAI obtains the right ascension of the CEO for a

given date in one of two ways. In high-accuracy mode, the values are read and interpolated from an external file (output from a numerical integration), which covers years 1700 to 2300. In low-accuracy mode, the values are obtained from a 16-term series valid from 1850 to 2150. NOVAS will generate an error message if you request dates outside these ranges or if it cannot find the external file (in high-accuracy mode). The path/name of the external file, its type (sequential or direct-access), and the logical unit number on which it is to be read can be specified in a call to CEOFIL, which obviously must precede any CEO-based computation. If CEOFIL is not called, NOVAS will assume that there is a formatted sequential file named CEO_RA.TXT in the current directory (folder) and will read it on logical unit 24. A copy of CEO_RA.TXT is available in the NOVAS directory (7.5 Mbytes), along with a utility program called CEO_file.f to convert it to a binary direct-access file (2.9 Mbytes) if desired. Existing NOVAS programs, unless specifically modified, will not use the CEO-based paradigm for Earth rotation and will not need any version of this file.

Appendix

Changes to NOVAS – From Version F2.0 (1998) to Version F2.9 (2004)

New Subroutines

NU2000A – from IERS (Wallace), evaluates IAU 2000A nutation series (nutation only).

NU2000K – modification of NU2000A, evaluates truncated version of full IAU 2000 A. More accurate than IAU 2000 B series. Also uses a consistent set of expressions for the fundamental arguments, those of Simon et al. (1994). Accuracy: about 0.1 mas for $\Delta\psi$, about 0.04 mas for $\Delta\epsilon$ and $\Delta\psi \sin \epsilon$.

EECT2000 – from IERS (Wallace), evaluates 34-term series for “complementary terms” in equation of the equinoxes.

EROT – evaluates the Earth rotation angle θ .

FRAME – sets up frame tie matrix and transforms vector from dynamical mean J2000.0 system to ICRS, or vice versa. FRAME implements a first-order matrix with second-order corrections to the diagonal elements, patterned after what is given in the Hilton and Hohenkerk (2004) A&A paper. Given the smallness of the angles involved and their uncertainties, this is quite adequate.

PLACE – New, general-purpose subroutine for computing apparent, topocentric, virtual, astrometric, etc., places of stars and planets. All substantive code for performing these calculations has been moved from APSTAR, TPSTAR, APPLAN, etc., into PLACE. In the call to PLACE, the object requested is specified by name, using a character argument, e.g., ‘SUN’, ‘MOON’, ‘JUPITER’, ‘STAR’, etc. The type of place requested is specified by two input codes, one indicating the location of the observer and the other indicating the coordinate system of the output positions. APSTAR, TPSTAR, APPLAN, etc., now are just “front-ends” to PLACE.

SETVEC – stores the last-computed celestial position vector.

GETVEC – allows the user to retrieve the last-computed celestial position as a unit vector.

IDSS – returns the planet number of a specified solar system body, to be used in calls to SOLSYS, APSTAR, APPLAN, etc. Actually a FUNCTION. The solar system body is specified by its name (all upper case letters) in the character variable that is the single input argument. For example, IDSS(‘EARTH’) = 3 (usually). A version of IDSS must now be packaged with each version of SOLSYS.

ASTCON – provides values of astronomical constants.

SETDT – allows user specification of ΔT (=TT–UT1) value in seconds. The ΔT value set here is used both in SIDTIM and TERCEL and, in certain circumstances, in PLACE.

GETDT – retrieves ΔT value (in days) previously specified via SETDT (in seconds).

ICRSEQ – transforms ICRS RA & Dec to RA & Dec on mean or true equator of date. For true equator of date, either the true equinox or the CEO can be specified as the origin of right ascension.

EQECL – converts equatorial RA & Dec to ecliptic longitude and latitude (both input and output are dynamical coordinates, either mean or true).

EQGAL – converts ICRS RA & Dec to galactic longitude and latitude.

DLIGHT – evaluates the difference in light-time to a star between the solar system barycenter and the Earth.

GRVDEF – replacement for SUNFLD that supervises the evaluation of gravitational deflection of light due to the Sun, Jupiter, and other solar system bodies. Calls new subroutine GRVD to do the deflection calculation for each body.

GEOPOS – called from PLACE to compute the geocentric position and velocity vectors of an observer on or above the surface of the Earth.

LITTIM – called from PLACE to antedate the position of a solar system body for light-time.

LIMANG – evaluates where an observed object is with respect to the Earth’s limb (horizon), given the geocentric position vectors of the observer and the object. PLACE calls LIMANG for the topocentric cases in deciding whether to include the gravitational deflection of light due to the Earth itself.

CEORA – returns the value of the true right ascension of the CEO for a given TDB Julian date.

CEORAI – returns the value of the ICRS right ascension of the CEO for a given TDB Julian date.

CEORD – for high-accuracy calculations, reads and returns a set of values of the ICRS right ascension of the CEO, near a given TDB Julian date, from an external file (either formatted sequential or binary direct-access).

CEOFIL – allows the specification of the external file of CEO right ascensions that CEORD reads.

CEOBAS – returns orthonormal basis vectors for non-rotating equatorial (intermediate) system.

SETMOD – sets method/accuracy mode for Earth rotation calculations.

GETMOD – retrieves method/accuracy mode for Earth rotation calculations.

EQINOX – specifies that equinox-based method is to be used for Earth rotation calculations.

CEOTEQ – specifies that the CEO-based method is to be used for Earth rotation calculations.

HIACC – specifies that high-accuracy ($\sim 1 \mu\text{as}$) algorithms are to be used for Earth rotation calculations.

LOACC – specifies that low-accuracy ($\sim 0.1 \text{ mas}$) algorithms are to be used for Earth rotation calculations.

RESUME – reverts to method/accuracy mode used prior to latest change (by one of the above subroutines).

Changes to Existing (NOVAS Version F2.0) Calling Sequences

APSTAR, VPSTAR, ASSTAR, CATRAN, GETHIP, VECTRS – proper motion units (in both RA and Dec) changed to milliarcseconds/year (pm in RA includes $\cos \delta$ factor), parallax units changed to milliarcseconds.

TPSTAR, TPPLAN, LPSTAR, LPPLAN – the user’s option to specify the input time argument as apparent sidereal time in hours is now discouraged; specifying the corresponding UT1 Julian date is now recommended. Sidereal time input is still supported but might not be in future NOVAS releases.

PRECES, CATRAN – one of the input epochs must now be 2451545.0 (J2000.0). Can no longer do two arbitrary epochs (the new precession expressions are not as flexible as Newcomb’s or Lieske’s).

CATRAN – has two new transformation options: IT=4 rotates data from the dynamical equator and equinox of J2000.0 to the ICRS and IT=5 does the opposite rotation.

WOBBLE – Julian date argument added.

PNSW – name changed to TERCEL (TERrestrial-to-CElestial transformation). Input argument changed to UT1 Julian date in pair of double-precision words.

CELPOL – input corrections to pole position can now be either (ΔX , ΔY) or ($\Delta\Delta\psi$, $\Delta\Delta\epsilon$), the choice specified by a new input parameter. Units must now be in milliarcseconds.

SPIN – no longer specifically associated with sidereal time. Now applies a rotation about the current z-axis, with angle expressed in degrees.

SUNFLD – replaced by GRVDEF, a more general subroutine that evaluates the gravitational deflection of light due to several solar system bodies.

All of the high-level subroutines (PLACE, APSTAR, APPLAN, etc.) now assume that they are working with ICRS data; this goes for the input RA, Dec, and proper motion components for the star routines and the position and velocity vectors obtained from SOLSYS (e.g., from DE-405) in both the star and planet routines. VPSTAR, LPSTAR, VPPLAN, LPPLAN, ASSTAR, ASPLAN, and MPSTAR produce output positions in the ICRS.

Significant Internal Changes to Code

Common error conditions will now generate error messages sent to unit=* (standard output, usually the terminal screen). Each error message always begins with the name of the subroutine that produced it, and is a plain-English description of the problem.

All subroutines that need astronomical constants now call ASTCON to obtain the values they need on their first call. Those values are SAVED for use on subsequent calls. Those values are:

SPEED OF LIGHT IN METERS/SECOND — A DEFINING PHYSICAL CONSTANT:

$$c = 299,792,458$$

LIGHT-TIME FOR ONE ASTRONOMICAL UNIT IN TDB SECONDS, FROM DE-405:

$$a(\text{sec}) = 499.0047838061$$

SPEED OF LIGHT IN AU/DAY:

$$c(\text{AU/day}) = 86400 / a(\text{sec})$$

LENGTH OF ASTRONOMICAL UNIT IN METERS:

$$a = a(\text{sec}) \times c$$

HELIOCENTRIC GRAVITATIONAL CONSTANT IN METERS³/SECOND², FROM DE-405:

$$GS = 1.32712440017987 \times 10^{20}$$

GEOCENTRIC GRAVITATIONAL CONSTANT IN METERS³/SECOND², FROM DE-405:

$$GM = 3.98600433 \times 10^{14}$$

EQUATORIAL RADIUS OF EARTH IN METERS, FROM IERS CONVENTIONS (2003):

$$r_{\oplus} = 6,378,136.6$$

FLATTENING FACTOR OF EARTH, FROM IERS CONVENTIONS (2003):

$$f = 1 / 298.25642$$

NOMINAL MEAN ROTATIONAL ANGULAR VELOCITY OF EARTH, IN RADIANS/SECOND, FROM IERS CONVENTIONS (2003):

$$\omega = 7.2921150 \times 10^{-5}$$

RECIPROCAL MASSES (SUN MASS/BODY MASS) FOR SOLAR SYSTEM BODIES

$$\text{SUN} = 1$$

$$\text{MOON} = 27,068,700.387534$$

$$\text{MERCURY} = 6,023,600$$

$$\text{VENUS} = 408,523.71$$

EARTH = 332,946.050895
MARS = 3,098,708
JUPITER = 1,047.3486
SATURN = 3,497.898
URANUS = 22,902.98
NEPTUNE = 19,412.24
PLUTO = 135,200,000
EARTH-MOON BARYCENTER = 328,900.561400

DE-405 values are used for many of these, which are in TDB (T_{eph}) units. NOVAS output is practically insensitive to changes in low-order digits of the above constants; they are mostly used for relatively small corrections, such as the gravitational deflection of light. Probably the light-time for 1 AU is the most important, because it is used for the light-time correction. The constants that really matter in NOVAS are the coefficients to the series expansions in the individual subroutines, i.e., the constants that are embedded in the models for precession, nutation, etc.

APSTAR, TPSTAR, APPLAN, TPPLAN, VPSTAR, LPSTAR, VPPLAN, LPPLAN, ASSTAR, ASPLAN – now are simply “front-ends” to specific calls to PLACE. All substantive apparent place calculations of various kinds are now done only in PLACE. The following changes in the basic algorithms were made:

- (1) Calls to FRAME were added in appropriate places to transform between the ICRS and the dynamical system.
- (2) In updating a star’s position for proper motion, there is now a correction to the epoch of interest for the difference in light-time between the solar system barycenter (the reference point for the input catalog data) and the Earth itself. (This affects only stars with the greatest proper motions, and then only at the 0.1 mas level). Uses the new subroutine DLIGHT to compute the epoch offset.
- (3) The “Doppler factor”, k , is included in the computation of stars’ space motion vectors (see note on VECTRS).
- (4) Modifications were made related to the change in gravitational deflection algorithms from SUNFLD to the more-general GRVDEF (see note on GRVDEF).
- (5) Code has been introduced that allows a place to be expressed in the celestial intermediate system (equator of date with CEO as right ascension origin).
- (6) Code has been added that allows the input of an observer’s instantaneous geocentric position and velocity vectors (with respect to the true equator and equinox of date) for a topocentric place calculation; this is included to support satellite observations.

TERCEL, SOLSYS – calls to FRAME added at appropriate places. (In SOLSYS, the call to FRAME is commented out for DE-405 and later JPL ephemerides, since DE-405 is in ICRS.)

CATRAN, GETHIP, VECTRS – code adjusted for new proper motion and parallax units.

CATRAN – code added to call FRAME for new IT=4 and IT=5 options that rotate data between dynamical J2000.0 system and ICRS.

VECTRS, CATRAN – $1/(\sin(\text{parallax}))$ now used to compute distance rather than $1/\text{parallax}$; an inconsequential change, just to make the expression formally correct. Also, the “Doppler Factor”, k , mentioned in the Hipparcos documentation and other papers, is now applied in computing the space-motion vector. The change in the units of proper motion and parallax is also implemented here.

SIDTIM – returns value of sidereal time, either mean or apparent. Internally can work by either of two methods, set by previous call to SETMOD, EQINOX, or CEOTEO:

Equinox-based method: Evaluates expression for sidereal time given in IERS Conventions (2003), Chapter 5, eq. (35). For apparent sidereal time, last three terms are considered part of equation of the equinoxes, obtained from ETILT. The Earth rotation angle θ is obtained from EROT.

CEO-based method: Obtains sidereal time from eq. (6) given in Kaplan (2003), based on the position of the equinox in the celestial intermediate system. The orthonormal basis of the celestial intermediate system is obtained from CEOBAS and the Earth rotation angle θ is obtained from EROT. Mean sidereal time, when requested, is obtained by *subtracting* the equation of the equinoxes, obtained from ETILT.

In either method, SIDTIM/EROT evaluates θ using the input UT1 epoch, but other components of sidereal time are evaluated using TDB (set equal to TT), with $TT=UT1+\Delta T$. Default value is $\Delta T=64$ sec, applicable at or near 2000; for highest precision applications, ΔT value can be set via prior call to SETDT.

TERCEL – performs the terrestrial-to-celestial transformation on a given vector, i.e., the total rotation from the ITRS to the ICRS. Internally can work by either of two methods, set by previous call to SETMOD, EQINOX, or CEOTEO:

Equinox-based method: Evaluates the old-style transformation as per previous subroutine PNSW, but with a call to FRAME added at the end to put final vector in ICRS. Uses apparent sidereal time, obtained from SIDTIM.

CEO-based method: Performs the transformation of eq. (4) given in Kaplan (2003), based on the celestial intermediate system. The orthonormal basis of the celestial intermediate system is obtained from CEOBAS and the Earth rotation angle θ is obtained from EROT.

In either method, the “fast angle” (rotation about z axis) is evaluated using the input UT1 epoch, but other components of the transformation are evaluated using TDB (set equal to TT), with $TT=UT1+\Delta T$. Default value is $\Delta T=64$ sec, applicable at or near 2000; for highest precision applications, ΔT value can be set via prior call to SETDT.

ETILT – now evaluates a more complete series for the complementary terms in the equation of the equinoxes (formerly just the two largest terms). Internally works in either high- or low-accuracy mode, set by previous call to SETMOD, HIACC, or LOACC:

High-accuracy mode: Obtains the sum of the terms from IERS function EECT2000.

Low-accuracy mode: Obtains the sum of the terms from a 9-term internal series.

ETILT uses the expression for the mean obliquity from Lieske et al. (1977), with a rate adjustment as per IERS Conventions (2003), Chapter 5, eqs. (31)-(32).

PRECES – now evaluates precession-angle polynomials from IERS Conventions (2003), Chapter 5, eq. (33) (with extra significant digits added to coefficient values). Some code changes made to ensure reversibility of transformation (to/from J2000.0).

NOD – now just calls either of the nutation subroutines, NU2000A (from the IERS) or NU2000K (a reduced-accuracy version of NU2000A), to do the hard work; does not contain nutation series itself. Which of the two nutation subroutines is called depends on whether high-accuracy or low-accuracy mode has been chosen for Earth rotation calculations (see new subroutines SETMOD, LOACC, HIACC).

FUNARG – now evaluates Simon et al. (1994) expressions for the fundamental solar and lunar arguments. However, IERS subroutine NU2000A, that evaluates the full nutation series, develops its fundamental arguments internally (a mixed bag of expressions).

WOBBLE – very tiny (inconsequential for most applications) rotation about z axis added to matrix to correct ITRS longitude origin to TEO, using recently published approximation to TEO longitude as a function of time (which requires the new time argument to this subroutine). Essentially, this changes W rotation to W'. Also changed matrix element expressions from first-order approximations to exact expressions for increased precision.

Other Internal Code Changes

Many minor changes have been made in the code. Obviously many of the comment statements had to be revised, and others added, too numerous to try to list. Some of the code is now more Fortran-77-like and less Fortran-66-like, especially in the subroutines in which other changes had to be made; a uniform scrub was not done. NOVAS still has plenty of ancient-looking code, it's still all-caps, and there are still some GO TOs. On the other hand, since NOVAS is mostly computational, flowing top-to-bottom, without any complicated logic, it hardly matters.

Some variable names were changed. For example, the variable PI in some subroutines was used for the parallax and not the mathematical constant $\pi=3.14159\dots$, which seemed nutty. In these cases, the variable name is now PX. The input (catalog) RA and Dec for many subroutines had been named RAM and DECM, the M indicating “mean”; these are now RAI and DECI, the I indicating “ICRS”. Many similar trivial changes have been made.

Unfinished Business

The precession algorithm used in NOVAS version F2.9, from the IERS Conventions (2003), is a temporary algorithm. The IAU Working Group on Precession and the Ecliptic has been charged with recommending a dynamically consistent precession theory for use as the new IAU standard. The working group has evaluated several candidate theories and seems to be moving toward endorsing the Capitaine, et al. (2003) precession, also known as P03 — but with a correction. The working group's final recommendation is expected in mid-to-late 2004. NOVAS version F3.0 will incorporate the new IAU precession theory as recommended by the working group, but will not be made public until the working group's final report has been accepted for publication (this whole topic has had too many examples of last-minute “adjustments”). Meanwhile, NOVAS version F2.9 will be used for the U.S. sections of the 2006 *Astronomical Almanac*.